FORWARD

℄ FRONT SPAR

FORM TO WING CONTOUR (STA. 2)

ON INSTALLATION DRILL (HOLES) 2 HOLES (MARKED ⊕) SEE LOCATION NOTE

CEMENT NEOPRENE TO WELDED ASSY.

STAGGERED SPOTWELD

℄ STRUT

ALL RADII ARE TO INNER FACE OF SKIN

AN526-DD1032-7
AC364-DF1032
AN935-D3
AN960-D10

DRILL END HOLE IN -11 L&R (OSA) ONLY ON ASSY. TO WING

L.H. SHOWN - R.H. OPPOSITE
6" = 1'-0"

SECTION A-A
12" = 1'-0"

INTERSECTION LINE

DRILL #28 (.140) 8 HOLES (MARKED ⊕) AT INST. ON WING

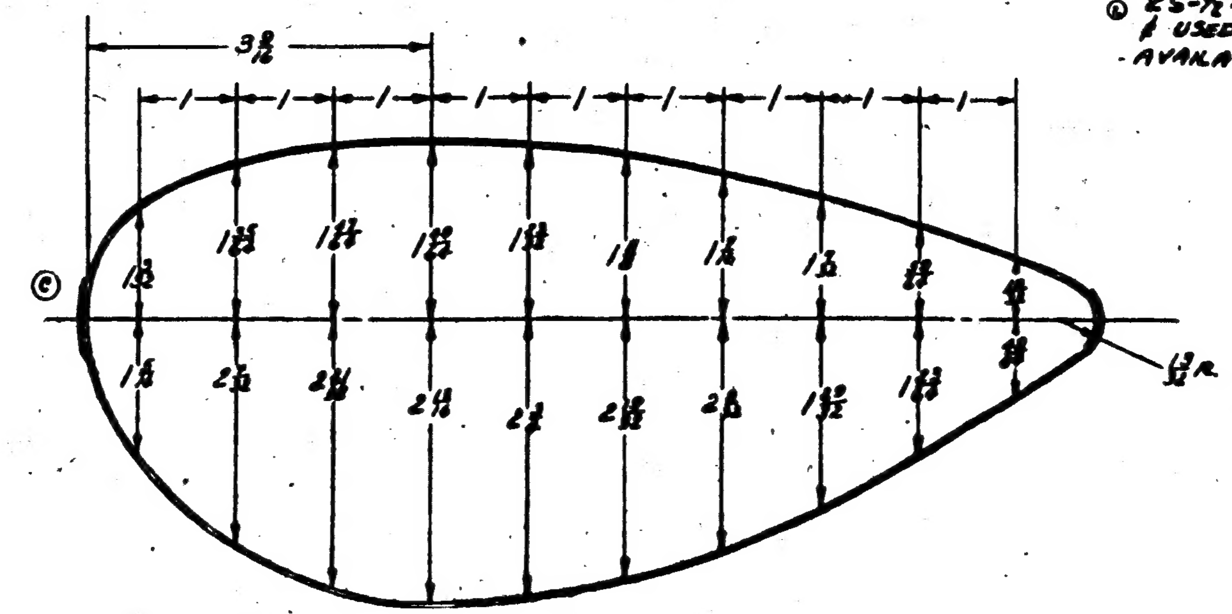
NOTE: LOCATE FAIRING HOLES FROM HOLES IN WING. SEE 28W5005 & 28W5016

VIEW D-D
6" = 1'-0"

SECT. E-E
6" = 1'-0"

Note: ① Contour Same As 28W075
② 2S-½H MATL MAY BE ANNEALED & USED FOR -8, -11 & -12 UNTIL 2SO IS AVAILABLE

SECT. B-B
12" = 1'-0"



SECT. C-C
12" = 1'-0"

P. GREEN PRIME

FINISH CODE	
Z	ANODIC TREATMENT
P	P-27 PRIMER
AP	ALUMINIZED PRIMER
C	CADMIUM PLATE

SCALE NOTED

FINISH NONE   Z-P-2 6P

TITLE FAIRING - WING FRONT STRUT - UPPER

CONSOLIDATED AIRCRAFT CORP.

J 28W5075

H 508